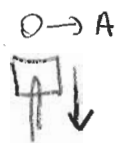
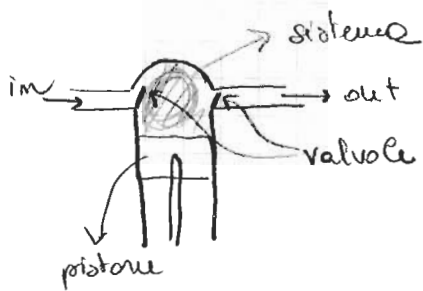
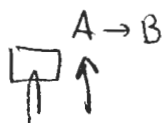


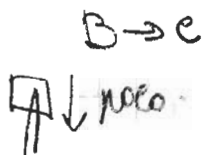
Esercizio sul motore diesel



0) entro da "in" aria
 P cost. e il pistone scende



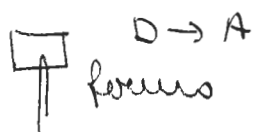
1) compressione adiabatica



2) in B inietto il carburante
 \Rightarrow aria + carburante si espande
 a pressione costante



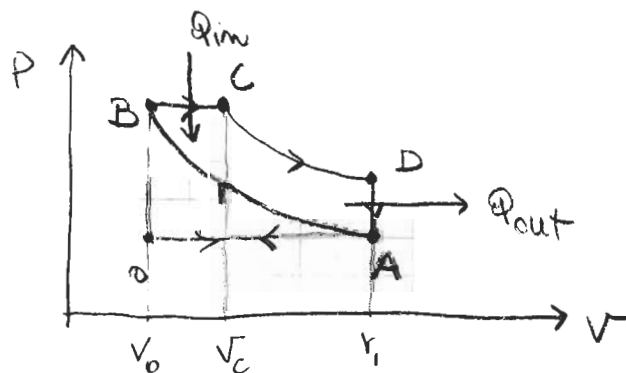
3) alta temp. delle miscela
 causa combustione \Rightarrow scoppio
 e espansione adiabatica



4) "out" aperte e miscela espulsa
 molto rapidamente



5) miscela espulsa con
 pistone che sale



$$\Delta U = 0 \Rightarrow \Delta L = \Delta Q = Q_{Be} + Q_{DA}$$

$$Q_{Be} = m C_p (T_c - T_B) \Rightarrow$$

$$Q_{DA} = m C_v (T_A - T_D)$$

$$\eta = \frac{AL}{Q_{Be}} = \frac{Q_{ec} + Q_{DA}}{Q_{Be}} = 1 + \frac{c_p}{c_f} \frac{T_A - T_D}{T_c - T_B}$$

$$\eta = 1 - \frac{1}{\gamma} \frac{T_D - T_A}{T_c - T_B}$$

A → B $T_A V_A^{\gamma-1} = T_B V_B^{\gamma-1}$ $T_A = T_B \left(\frac{V_B}{V_A}\right)^{\gamma-1}$

c → D $T_c V_c^{\gamma-1} = T_D V_D^{\gamma-1}$ $T_D = T_c \left(\frac{V_c}{V_D}\right)^{\gamma-1}$

$$\eta = 1 - \frac{1}{\gamma} \frac{T_c \left(\frac{V_c}{V_D}\right)^{\gamma-1} - T_B \left(\frac{V_B}{V_A}\right)^{\gamma-1}}{T_c - T_B}$$

Facciamo qualche conto: trovare la potenza fornita da un motore diesel con 4 cilindri, 2 L di cilindrata, 3000 giri al minuto, $r = \frac{V_A}{V_B} = 22$

e $\frac{V_c}{V_B} = \lambda = 2$ $T_A = 300 \text{ K}$
 $P_A = 10^5 \text{ Pa}$

V_A e V_B sono formule del pb del motore a scoppio ⇒

$$\left\{ \begin{aligned} V_A &= \frac{r}{r-1} \Delta V = 0.523 \cdot 10^{-3} \text{ m}^3 \\ V_B &= \frac{\Delta V}{r-1} = 0.238 \cdot 10^{-4} \text{ m}^3 \end{aligned} \right.$$

$$m = \frac{P_A V_A}{RT_A} = 0.02101 \approx 0.609 \text{ g di aria}$$

A → B adiabática

$$P_B = P_A r^\gamma = 7.57 \cdot 10^6 \text{ Pa}$$

$$T_B = 1033 \text{ K}$$

B → C isobara

$$P_C = P_B$$

$$V_C = x V_B$$

$$\Rightarrow T_C = 2T_B = 2066 \text{ K}$$

C → D adiabática

$$V_D = V_A$$

$$P_D V_A^\gamma = P_C V_C^\gamma$$

$$P_D = P_C \left(\frac{V_C}{V_A} \right)^\gamma = P_C \left(\frac{x}{r} \right)^\gamma = 2.64 \cdot 10^5 \text{ Pa}$$

$$\Rightarrow T_D = \frac{P_D V_A}{nR} = 790 \text{ K}$$

Rendimiento

$$\eta = 1 - \frac{1}{r} \frac{T_D - T_A}{T_C - T_B} = 0.66 \Rightarrow 66\%$$

$$P_{\text{tot}} = 4 \times \frac{2000}{60} \times \Delta L \frac{1}{2}$$

$$\Delta L = n c_p (T_C - T_B) + m c_v (T_A - T_D) = 845 \text{ J}$$

$$\Rightarrow P_{\text{tot}} = 56.3 \text{ kW.}$$